Mars Global Surveyor

Thermal Emission Spectrometer

Data Processing User's Guide

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### 1.0 OVERVIEW

INSTRUMENT DESCRIPTION
The Thermal Emission Spectrometer (TES) investigation is designed to study the surface and atmosphere of Mars using thermal infrared (IR) spectroscopy, together with broadband thermal and solar reflectance radiometry. The specific objectives of the TES experiment are: (i) to determine and map the composition of surface minerals, rocks, and ices; (ii) to study the composition, particle size, and spatial and temporal distribution of atmospheric dust; (iii) to locate water-ice and CO 2 condensate clouds and determine their temperature, height, and condensate abundance; (iv) to study the growth, retreat, and total energy balance of the polar cap deposits; (v) to measure the thermophysical properties of the martian surface materials; and (vi) to characterize the thermal structure and dynamics of the atmosphere. A complete description of the TES instrument is given in (Christensen et al., 1992).

The TES instrument consists of three sub-sections, the primary one being a Michelson interferometer that produces spectra from 1700 to $200 \mathrm{~cm}-1$ ( $\sim 6$ to 50 $\mu \mathrm{m}$ ), at a spectral sampling of either $\sim 5$ or $\sim 10 \mathrm{~cm}-1$. The instrument cycle time, including collection of the interferogram, mirror flyback, and electronic reset, is 2 sec for $10 \mathrm{~cm}-1$ ("single scan") operation, and 4 sec for $5 \mathrm{~cm}-1$ ("double scan") operation. The interferometer includes a visible interferometer with a monochromatic source that is used to generate fringes which control the linear drive servo and determine position in the interferogram. This system uses two redundant neon lamps that produce an emission line at 703.2 nm for fringe generation and a continuum that is used for a quasi-white-light source for determination of zero path difference.

TheTES instrument returns 143 points in single-scan or 286 points in double-scan mode. The starting spectral sample point can be determined by ground command. In single-scan mode the default PROM sequence for Detector 2 begins at $148.6 \mathrm{~cm}-1$ and ends at $1655.9 \mathrm{~cm}-1$. This spectral range was used throughout the aerobraking and Science Phasing Orbits. However, the first five spectra samples in single-scan mode (first 10 in double-scan mode) have very low instrument response and a very low signal-to-noise ratio. Therefore, beginning with the mapping orbits the starting sample in single-scan will be changed to $201.6 \mathrm{~cm}-1$ (Det. 2) with an ending sample of $1708.9 \mathrm{~cm}-1$. The single-scan data are stored in a 148 -point array beginning at $148.6 \mathrm{~cm}-1$ (Det. 2) in which either the first five (mapping phase) or last five (aerobraking phase) samples are set to zero or null. The double-scan data are stored in a 296-point array with corresponding offsets and null values. The wavenumber positions for each detector from sample 1 to 148 (single scan) or sample 1 to 296 (double scan) are given in Table A1.

The finite size and off-axis position of the six detectors results in selfapodization and a spectral shift that is a function of both distance from the axis and optical frequency. The resulting full-width half-maximum (FWHM) value is $\sim 12.5 \mathrm{~cm}-1$ for $10 \mathrm{~cm}-1$ sampling at $200 \mathrm{~cm}-1$ and $15.4 \mathrm{~cm}-1$ at $1650 \mathrm{~cm}-1$. For the corner detectors and at the highest frequency (shortest wavelength) there is a significant departure from the ideal line width, giving a worst-case degradation of a FWHM of $\sim 24 \mathrm{~cm}-1$. Because all of the response functions have the same area there is no loss in signal when viewing a smooth continuum scene like Mars. However, there will be a slight loss in contrast of narrow spectral features due to broadening of the spectral width. Because the self-apodization is considerable, the data are used without further apodization. Separate fast fourier transform (FFT) algorithms are used for the center and edge detectors in order to partially compensate for the different spectral shifts introduced into these detectors. These offsets are discussed in Section 2.4.

A pointing mirror capable of rotating $360^{\circ}$ provides views to space, both limbs, and to internal, full-aperture thermal and visible calibration targets, as well
as image motion compensation. In addition to the spectrometer, the instrument has bore sighted bolometric thermal radiance ( 4.5 to $\sim 100 \mu \mathrm{~m}$ ) and solar reflectance ( 0.3 to $2.7 \mu \mathrm{~m}$ ) channels. Each instrument sub-section has six instantaneous fields of view (IFOV) of $\sim 8.5$ mrad that provide a contiguous strip three elements wide with a spatial resolution designed to be 3 km from the final MGS mapping orbit altitude of 350 km . The outputs from all TES channels are digitized at 16 bits, processed, and formatted before being sent to the spacecraft Payload Data Subsystem (PDS). The outputs of the interferometer receive the following processing within the instrument before transfer to the PDS: 1) selectable apodization; 2) Fast Fourier Transformation (FFT) of data from all six interferometer channels; 3) correction for gain and offsets; 4) data editing and aggregation; 5) data compression; and 6) formatting for the PDS.

A separate 1.5 cm diameter reflecting telescope, collimated with the main telescope and using the same pointing mirror, is used for the thermal and visible bolometer channels. These channels have similar $3 \times 2$ arrays of detectors, that are bore sighted with the spectrometer array. The optical system consists of a single off-axis paraboloidal mirror operating at $\mathrm{f} / 8$. A reflecting resonant fork chopper operating at 30 Hz is used to separate the solar reflectance and thermal emission bands.

### 2.0 SPECTROMETER CALIBRATION

### 2.1 SPECTROMETER ALGORITHM OVERVIEW

The measured spectra can be characterized at each wavenumber by the equation:
$\mathrm{Vt}=(\mathrm{Rt}-\mathrm{Ri}) * \mathrm{IRF}$
where,

Vt is the voltage generated by the TES looking at a target
Rt is the radiance of the target
Ri is the radiance of the instrument
IRF is the instrument response function
The radiance of the target can be determined from the above equation once the instrument radiance and the response function are known. These parameters are determined using observations of space and the internal reference surface at planned time intervals. These observations give two equations of the form:

$$
\begin{aligned}
& \mathrm{Vr}=(\mathrm{Rr}-\mathrm{Ri}) * \mathrm{IRF} \\
& \mathrm{Vs}=(\mathrm{Rs}-\mathrm{Ri}) * \mathrm{IRF}
\end{aligned}
$$

where Vr and Vs are the measured voltages viewing space and reference
respectively, Rr is derived from the measured temperature of the reference surface, and Rs is the radiance of space ( $\sim 0 \mathrm{~W} \mathrm{~cm}-2$ str- $1 / \mathrm{cm}-1$ ). These equations can be solved for the two unknown values, Ri and IRF, giving:

$$
\begin{aligned}
& \mathrm{Ri}=(\mathrm{Vs} * \mathrm{Rr}-\mathrm{Vr} * \mathrm{Rs}) /(\mathrm{Vs}-\mathrm{Vr}) \\
& \mathrm{IRF}=\mathrm{Vr} /(\mathrm{Rr}-\mathrm{Ri})
\end{aligned}
$$

Or the equivalent:

$$
\mathrm{IRF}=\mathrm{Vs} /(\mathrm{Rs}-\mathrm{Ri})
$$

These computed values are then used to compute the radiance of the planet using:

$$
R p=(V p / I R F)+R i
$$

### 2.2 SPECTROMETER ALGORITHM VERSION (V.002A)

The simultaneous determination of IRF and Ri requires Space (S) and Reference surface ( R ) observations spaced closely in time. Typically these are acquired as consecutive or interleaved observations that are termed "SR-pairs". The IRF is assumed to vary slowly, whereas Ri can vary throughout the orbit. Thus, the SRpairs are only acquired several times per orbit to determine IRF, whereas Space observations are acquired approximately every 3-5 minutes to determine Ri.

It is necessary for the calibration that the required subsets of all the parameters are also available. For example, the spectral values for the planet acquired from detector 5 can only be calibrated if all other parameters are also available for detector 5 . Similarly, single-scan planet observations are calibrated using single-scan S and R observations, and double-scan planet observations require double-scan S and R observations.

During the aerobraking and Science Phasing Orbits the thermal state of the TES was not stable. For example, during each spacecraft roll the Sun could directly illuminate the reference surface. Therefore, it was not possible to use longterm averages of IRF and Ri to reduce the noise level present in a single determination of these parameters. In this version of the algorithm the bounding values of IRF and Ri are simply interpolated to determine Rp. The instrument response was not averaged over multiple SR-pairs, nor was the Ri term smoothed to reduce noise.

The following sequence of operations was carried out for spectral calibration:

1) Read the data associated with all the observations under consideration
2) Find all of the single and double scan SR-pairs and Space observations (S) in the given set of observations.
3) At each SR-pair, compute the temperature of the instrument (Ti) and IRF. For each detector:
a) Average the voltage of all the Space observations having the same scan length. This is Vs.
b) Average the voltage of all the reference observations having the same scan length. This is Vr. Average the reference surface thermistor temperatures (aux_temp[1-3]) to find the average temperature of the reference surface for this SR-pair. This is Tr .
c) Compute the radiance of the reference surface $(\mathrm{Rr})$ at temperature Tr using the Planck blackbody radiance function.
d) Compute the radiance of space (Rs) at the temperature of space $(3 \mathrm{~K})$, using the Planck radiance function.
e) Compute the radiance of the instrument by substituting the calculated values in the equation:

$$
\mathrm{Ri}=(\mathrm{Vs} * \mathrm{Rr}-\mathrm{Vr} * \mathrm{Rs}) /(\mathrm{Vs}-\mathrm{Vr})
$$

f) Compute the instrument brightness temperature (Ti) at each spectral sample by inverting the blackbody radiance function with radiance Ri.
$\mathrm{g})$ Take the average of the instrument brightness temperatures from spectral samples 50 through 90 (single scan; samples 100-180 double scan), to determine a single best-fit value of Ti . This is the temperature of this particular detector.
h) Compute IRF using the equation:

$$
\mathrm{IRF}=\mathrm{Vs} /(\mathrm{Rs}-\mathrm{Ri})
$$

If IRF equals zero or infinity for a particular spectral sample, then average the two neighboring spectral samples to compute an IRF value for that spectral sample.
4) Store the computed values of IRF, Ri, and Ti into one packet, tag it as an SR-pair with its starting sclk_time and pool it among other similar packets for SR-pairs and Space observations in ascending order of their sclk_time. This pool is called the IRF-pool.
5) Replicate the first SR-pair as an additional SR-pair in the beginning of the given set of observations.
6) Replicate the last SR-pair as an additional SR-pair at the end of the given set of observations.
7) At each Space observation, compute Ti. For each detector:
a) Average the voltage of all the Space spectra in a given set of consecutive spectra having the same scan length. This is Vs.
b) Compute radiance of space (Rs) at temperature of space (3K) using the Planck radiance function.
c) Search in the IRF-pool to find the closest SR-pair in each direction. Interpolate over sclk_time between the Ti of the two bounding SR-pairs to compute an initial estimate of Ti at the sclk_time of this Space observation.
d) Interpolate over Ti between the IRF values at the Ti's of the two bounding SR pairs to compute the IRF at the estimated Ti of this Space observation.
e) Compute the radiance of the instrument using the value of IRF in the equation:

$$
\mathrm{Ri}=\mathrm{Rs}-(\mathrm{Vs} / \mathrm{IRF})
$$

f) Compute the instrument brightness temperature (Ti) at each spectral sample by inverting the blackbody radiance function with radiance Ri. g) Take the average of the instrument brightness temperatures from spectral samples 50 through 90 (single scan; samples 100 to 180 double scan) to determine a new value of Ti.
h) Store the final computed values of Ti and Ri for this Space observation into one packet. Tag this packet as an $S$ with its starting sclk_time and pool it in the IRF-pool in ascending order of its sclk_time.
8) At each planet observation, determine IRF and Ri and compute Rp. For each detector:
a) Interpolate over sclk_time between the Ti of the two bounding SR observations to compute Ti at this planet observation.
b) Interpolate over Ti between the IRF values at the Ti's of the two bounding SR observations to determine the IRF at this planet observation.
c) Interpolate over sclk_time between the Ri values of the two bounding SR or S points to determine Ri at this planet observation.
d) If this observation had a spectral mask other than full spectral resolution, average the Ri and IRF corresponding to the mask.
e) Use IRF and Ri to compute Rp using the equation:
$R p=(V p / I R F)+R i$.
9) Write the calibrated spectra to the database.

### 2.3 PRECISION AND ACCURACY

The TES spectrometer has a noise equivalent spectral radiance near $1.2 \times 10-8 \mathrm{~W}$ $\mathrm{cm}-2$ str- $1 \mathrm{~cm}-1$. This corresponds to a signal-to-noise ratio (SNR) of 490 at $1000 \mathrm{~cm}-1$ ( 10 microns) viewing a 270K scene. Absolute radiometric accuracy was estimated from pre-launch data to be better than $4 \times 10-8 \mathrm{~W} \mathrm{~cm}-2$ str- $1 \mathrm{~cm}-1$. In flight deviations are discussed in Section 6.

### 2.4 WAVENUMBER SAMPLE POSITION AND SPECTRAL LINE SHAPE

In an ideal interferometer with an on-axis point detector, the spectral samples are uniformly distributed in wavenumber, and the full-width, half maximum (FWHM) of each sample is simply determined by the optical displacement of the Michelson
mirror. The TES uses a neon bulb with a line at 0.7032 microns in the visible interferometer to sample the IR interferometer. The ideal sample spacing of the interferometer is given by:

Sample spacing $=$ 1

$$
\begin{equation*}
(0.7032 \times 10-4 \mathrm{~cm}) * \mathrm{Npts} \tag{2.1}
\end{equation*}
$$

where Npts is the number of points in the FFT.
For a large detector, the two beams of the interferometer are not in phase over the entire areal extent of the detector, producing "self-apodization", or widening of the instrument line shape. In addition, the path length of the rays traveling to the off-axis portion of each detector is decreased relative to the optical axis rays by a factor of $\cos \phi$, where $\phi$ is the angle of the offaxis ray. As a result, the mirror must move farther to produce interference of the off-axis rays, producing a shift of the center frequency of each spectral sample to a higher apparent wavelength (lower wavenumber) than its true spectral position. All six detectors are offset from the optical axis, producing separate shifts in the spectral line position, shape, and modulation efficiency of each detector.

The TES flight software processes the interferogram data with prime factors FFTs that use a different number of points for the center and edge detectors respectively. These FFT's were selected to produce a slightly different spacing that partially compensates for the different spectral offsets due to selfapodization between the edge and center detectors. The number of points and sample spacing is given in Table 2-1.

Table 2-1.

Npts in FFT
Edge Detectors (1,3,4,6) Center Detectors (2,5)
Single Scan Double Scan Single Scan Double Scan
Sample Spacing
$1350 \quad 2700$
13442688
Sample 1 Position
(ideal)
$10.53 \mathrm{~cm}-1 \quad 5.267 \mathrm{~cm}-1 \quad 10.58 \mathrm{~cm}-1 \quad 5.290 \mathrm{~cm}-1$
$147.47 \mathrm{~cm}-1 \quad 147.47 \mathrm{~cm}-1 \quad 148.13 \mathrm{~cm}-1 \quad 148.13 \mathrm{~cm}-1$
Sample 148 (single;
296 double) Position
(ideal) $\quad 1695.95 \mathrm{~cm}-1 \quad 1701.22 \mathrm{~cm}-1 \quad 1703.52 \mathrm{~cm}-1 \quad 1708.81 \mathrm{~cm}-1$
(Note: the sample spacing used to compute the sample position of the archived data is computed in full digital precision using 0.7032 microns in Eq. 2.1.)

A numerical model has been developed by Co-Investigator Stillman Chase to model the self-apodization effects and to determine the true spectral position, FWHM, and spectral line shape of each sample. Interferogram data of Mars were
collected immediately after Mars orbit insertion, and the atmospheric CO2 data were used to verify Chase's model of line shape and position. Because the focal plane is symmetric in the cross-track direction (e.g. detectors 1 and 3 are symmetrically located relative to the optical axis), the position and FWHM are identical for detector pairs 1 and 3 and detector pairs 4 and 6 . The sample position offset was calculated for each detector, taking into account the actual prime factor FFT used for each detector. Examples of the offset and the actual sample position calculated with this offset and the actual prime factors FFT used for the double-scan samples 1 and 296 are given in Table 2-2. The full set of sample offsets for each detector are tabulated in Table A1 in the Appendix. The sample positions are identical for single and double scans. The modeled full-width half-maximum values for double scan observations are tabulated in Table A2. The full-width half-maximum values for single scan data are twice the double scan values.

Table 2-2. Double Scan Self-Apodization

Edge Detectors $(1,3)$ Center Detector (2)
Sample 1 Self-Apodization
Offset
$1.19 \mathrm{~cm}-1$
$0.44 \mathrm{~cm}-1$
Sample 296 Self-Apodization
Offset
$14.00 \mathrm{~cm}-1$
$5.45 \mathrm{~cm}-1$
Sample 1 (actual)
$148.66 \mathrm{~cm}-1$
Sample 296 (actual)
$1715.22 \mathrm{~cm}-1$ $148.57 \mathrm{~cm}-1$ $1714.26 \mathrm{~cm}-1$

### 3.0 VISIBLE BOLOMETER CALIBRATION

### 3.1 VISIBLE BOLOMETER ALGORITHM OVERVIEW

The in-flight calibration of the TES visible bolometer is performed for each detector in the following stages.

1) Use observations of the internal TES reference calibration lamp and space to determine the instrument response function (IRF) and the zero-level radiance (background).
2) Convert each target observation to calibrated radiance using the IRF and background.
3) Compute the Lambert albedo using the calibrated radiance, the Sun-Mars distance, and the incidence angle.

### 3.2 VISIBLE BOLOMETER ALGORITHM VERSION V.002A

1)Read the data associated with all the observations to be calibrated. For each observation this includes:

| sclk_time | Spacecraft Clock Time |
| :--- | :--- |
| pnt_view | Pointing Angle View |
| det_mask | Detector mask |
| scan_len | Scan length |
| solar_distance | Solar distance |
| aux_temps[1-3] | Temperature of the reference surface |
| temps[1] | Temperature of the detector array |
| detector | Detector Number |
| vbol | Raw voltages from visible bolometer |
| incidence | Solar incidence angle at the target |
| latitude | Latitude of the target |

2) Separate the data into single and double scan modes. Each mode is calibrated separately.
3) Sort the data on ascending sclk_time.
4) Find all distinct groups of space observations (S) in the given set of observations. To improve on noise-reduction, only space observations with three or more consecutive space spectra are used; other space views with fewer than three space spectra are discarded.

At each set of Space observations for each detector:
a) Average the visible bolometer voltage (vbol) of all space observations in this set. This is called the background.
b)Store the background value with the beginning sclk_time of this set in one packet. Tag it as an S, and pool it among similar packets in ascending order of its sclk_time.
5) Find all distinct groups of consecutive visible bolometer reference observations (REFAn) within the given set of observations, where n refers to the calibration lamp number 1 or 2.
At each REFAn for each detector:
a) Compute the IRF for each internal lamp observation:
i) Average the vbol for all consecutive observations within this set of lamp observations.
ii) Correct vbol for the background signal by subtracting the background from the nearest Space observation to give the lamp_voltage.
iii) Average the temperature of the detector array (temps[1]) to yield detector temperature (det_temp) in degrees C .
iv) Average the three reference surface temperatures (aux_temps[1-

3]) to yield lamp temperature (lamp_temp).
v) Choose the lamp absolute radiance at 28.2 C (lamp_absolute) for this lamp and scan length (single or double) using the values in the TES Calibration Report.
vi) Select the ¿RL/ぇT for the lamp that was observed.
vii) Compute the actual lamp radiance (RL)(TL)cal for each detector at the lamp temperature using lamp_absolute, the variation in lamp radiance ( RL ) with temperature ( $\mathrm{\imath RL} / \mathrm{i} \mathrm{TL}$, ) from the TES Calibration Report, and the difference (TL) between lamp_temp and the internal
lamp absolute calibration temperature (28.2 degrees). The lamp radiance equation is:
lamp_radiance $=$ lamp_absolute $+(\AA \mathrm{RL} / \AA \mathrm{T} *($ lamp_temp -28.2) $)$ )
vii) Compute IRF using the equation:

IRF = lamp_voltage / lamp_radiance
b) Store the IRF, the background voltage, and the detector temperature in one packet. Tag it as a REFAn packet and pool it among other REFAn packets in ascending order of their beginning sclk_time.
6) Replicate the first REFAn as an additional REFAn in the beginning of the given set.
7) Replicate the last REFAn as an additional REFAn at the end of the given set.
8) At each target (planet) observation, compute the calibrated visible bolometer radiance (cal_vbol). The visible bolometer response function must be corrected at each planet observation to account for changes in the detector temperature between the lamp and planet observations. In order to avoid discrete jumps at each lamp observation, it is necessary to interpolate IRF and the detector temperature between successive lamp views. This baseline IRF is then corrected to the actual IRF at each planet observation using $\downarrow \mathrm{i} / \mathrm{\Delta TD}$, and $\downarrow 2 \mathrm{f} / \mathrm{\imath TD} 2$, along with the coefficients in the TES Calibration Report and the detector temperature difference between the baseline lamp observations and the planet observations. This is done by:
a) Search the pool of REFAn observations to find the two bounding observations.
b) Search the pool of Space observations to find the two bounding observations.
For each detector:
i) Interpolate linearly on sclk_time between this observation and two bounding REFAn times to compute the baseline IRF.
ii) Interpolate linearly on sclk_time between this observation and two bounding REFAn times to get the baseline detector temperature and the background for this observation.
iii) Determine $\ddagger$ IRF/ $\grave{\mathrm{T}}$ and $\downarrow 2 \mathrm{IRF} / \grave{\mathrm{T}} 2$ using the calibration coefficients determined pre-launch and the equations: iIRF/ $\mathrm{\imath T}=3 * \mathrm{af} *($ detector_temp $) 2+2 * \mathrm{bf} *$ detector_temp +cf ̀2IRF/ $\mathrm{\imath T} 2=6 *$ af $*$ detector_temp $+2 * \mathrm{bf}$ where,
af = 'alpha' of the visual bolometer from the TES
Calibration Report.
bf = 'beta' of the visual bolometer from the TES Calibration Report. $\mathrm{cf}=$ 'chi' of the visual bolometer from the TES Calibration Report.
iv) Get the actual detector temperature (temps[1]) for this observation.
v) Calculate delta T by subtracting the baseline detector temperature value from the actual value.
vi) Correct IRF for this detector temperature value by the
equation:
Corr_IRF $=\mathrm{IRF}+\grave{\mathrm{IRF}} / \grave{\mathrm{t}} \mathrm{T} * \Delta \mathrm{~T}+\grave{\mathrm{t}} \mathrm{IRF} / \mathrm{dT} 2 * \Delta \mathrm{~T} 2 / 2$ !
vii) Compute cal_vbol using the equation:
cal_vbol $=($ vbol - background $) /$ Corr_IRF
9) Compute Lambert albedo.
a) Extract incidence angle and solar distance from the database.

Convert solar_distance to Astronomical Units.
b) Compute albedo using the equation:
lambert_alb = cal_vbol / ( Sun_absolute / solar_distance2)

* $\cos$ (incidence_angle))
where Sun_absolute is the solar radiance at 1 A.U. integrated over the TES visible bolometer relative spectral response, and is equal to $1.666 \times 10-2 \mathrm{~W}$ cm-2 str-1
Note: The $\cos$ (incidence_angle) in the denominator can lead to division by small numbers (including zero), generating highly inaccurate values for the albedo. To avoid this problem, the Lambert albedo is not computed for incidence angles >88 degrees.

10) Write cal_vbol and lambert_alb to the database.

### 3.3 PRECISION AND ACCURACY

The precision, zero-level offset, and absolute accuracy of the in-flight calibration was determined using data from cruise (test tes_c2a and tes_c9a) and orbits P3 through P460.

The in-flight precision (noise level) of the calibrated radiance measurements was determined using observations of deep space acquired away from Mars during spacecraft rolls prior to and after periapsis.. The internal lamp was not used because its temperature increases if left on for an extended period of time, which changes its brightness level. The data used were acquired on orbits P95 through P100 (no data were available for orbit P99) at a Mars-Sun distance of $2.068 * 10-8 \mathrm{~km}$ (1.382 A.U.). Only observations well away from Mars, selected by constraining the height of the tangent point of the observation to be $>2000 \mathrm{~km}$ above the martian surface, were included. The sigma values of the calibrated radiance of the space observations are given in Table 3-1.

Table 3-1

| Detector | Sigma (Radiance) <br> $(\mathrm{x} 10-6 \mathrm{~W} \mathrm{~cm}-2 \mathrm{str}-1)$ | Mean Zero-level Radiance <br> $(\mathrm{x} 10-6 \mathrm{~W} \mathrm{~cm}-2 \mathrm{str}-1)$ |
| :--- | :---: | :---: |
| 1 | 3.62 | 0.914 |
| 2 | 3.74 | 1.03 |
| 3 | 3.77 | 1.07 |
| 4 | 3.73 | 0.676 |
| 5 | 3.67 | 0.942 |
| 6 | 3.59 | 1.00 |

The 1 sigma variation in the zero-level radiance is $\sim 3.75 \times 10-6 \mathrm{~W} \mathrm{~cm}-2$ str- 1 for all six detectors. This value is consistent with the variation in the internal lamp brightness measured pre-flight ( $1-6 \times 10-6 \mathrm{~W} \mathrm{~cm}-2$ str-1; Table 4-6). A Lambertian surface with a reflectivity of 1.0 would have a radiance of 8.718 x10-3 W cm-2 str-1 at the Mars-Sun distance of these observations, measured at normal incidence angle. The 1 sigma precision of the visible bolometer calibrated radiance corresponds to a noise-equivalent delta reflectivity ( $\mathrm{NE} \Delta \mathrm{R}$ ) of 0.0004 , and is equivalent to an SNR of 2100 for a surface with unit reflectivity.

The zero-level radiance as a function of time is determined by the calibration algorithm using periodic observations of space and the internal lamps to correct for detector response and offset drifts. Table 3-1 gives the mean zero-level radiance of the space observations. This radiance is a factor of nearly four lower than the $1 \sigma$ variation of the data, indicating that there are no measurable systematic biases introduced into the data by incorrectly removing the variations in detector response and lamp brightness with time and temperature. In addition, no systematic offsets, trends, or discrete changes in value at space or lamp observations were observed in the calibrated radiance of space. From these data it is concluded that the calibration algorithm is accurately accounting for variations in detector response and lamp brightness with time and temperature at the noise level of the instrument. The $3 \sigma$ accuracy of the zero-level radiance is approximately $+/-1 * 10-5 \mathrm{~W} \mathrm{~cm}-2$ str- 1 for all six detectors, consistent with the values in Table 3-1.

The data given in Table 3-1 were acquired of a black target (space) with zero signal and therefore do not provide a measure of the true absolute calibration for bright surfaces. This can only be determined by observations of a bright source with known radiance. No surfaces of known brightness exist on Mars to verify the absolute radiance. In addition, because the internal calibration lamps are used in the calibration, they do not provide an independent test of the absolute radiance. However, it is possible to estimate changes in the lamp output with time by comparing the measured lamp voltage, corrected for background, with the pre-flight measurements as a function of lamp and detector temperature.

The pre-flight thermal vacuum tests (albm tests) and the in-flight data from cruise (tests tes_c2 and tes_c9, and orbits 12, 15, 95-98, 100, 222, and 460) indicate a $0-\sim 3 \%$ increase in the measured signal for detector temperatures of $\sim 10-15$ degrees C, and an increase of $\sim 3-6 \%$ near 0 C relative to the pre-flight measurements. This change can be due to a combination of: 1) a change in the alignment of the lamp relative to the detectors; 2) an increase in lamp 1 brightness; 3) a change in the chopper alignment or timing; or 4) an increase in detector response. Of these, a change in alignment is least likely because no decrease in lamp signal was observed for any detector. A possible change in lamp 1 brightness was investigated using the ratio of lamp 1 to lamp 2
for pre-flight and in-flight data. Lamp 2 was observed once during cruise (test tes_c9) and once in orbit early in the mission (orbit P59). The lamp ratio, adjusted for lamp temperature, is unchanged for detectors 3 and 6 , is $\sim 1 \%$ higher for detectors 1,4 , and 5 , and is $\sim 1 \%$ lower for detector 2 . The change in lamp 1 relative to lamp 2, averaged for all detectors, is $\sim 0.7 \%$, and is essentially constant with temperature. Based on experience at SBRS on the Galileo PPR instrument, the stability of these lamps is estimated to be $+/-0.5 \%$ on a long-term (years) basis and +/- $0.15 \%$ on a short term (hours) basis. The ratio of the two TES lamps is consistent with these stability values. Furthermore, both lamps would have to have increased in brightness to account for changes in the lamp 1 signal levels. It is therefore concluded that the changes in lamp 1 signal level are not associated with changes in lamp output.

It is more likely that either the detector response with temperature has varied in flight, which would account for both the variations between detectors and the relatively large changes over temperature, or that the chopper alignment or timing has changed slightly. Neither of these cases will affect the absolute calibration because the detector views both Mars and the lamps with the same chopper and detector characteristics. Indeed, the on-board calibration lamps are specifically intended to remove these effects. It is concluded that the absolute calibration is most likely $\sim 1 \%$ relative to the pre-flight calibration of the internal lamps. The relative accuracy from orbit P15 to P460 is $\sim 0.5 \%$.

### 4.0 THERMAL BOLOMETER CALIBRATION

### 4.1 THERMAL BOLOMETER ALGORITHM OVERVIEW

The measured integrated radiance can be characterized by the following equation:
$\mathrm{Vt}=(\mathrm{Rt}-\mathrm{Ri}) * \mathrm{IRF}$
where,

Vt is the voltage generated by the TES looking at a target
Rt is the integrated radiance of the target
Ri is the integrated radiance of the instrument
IRF is the instrument response function

The integrated radiance of the target is determined from the above equation once the instrument radiance and the response function are known. These parameters are determined using observations of space and the internal reference surface at planned time intervals. These observations give two equations of the form:

```
Vr=( Rr - Ri)* IRF
Vs=(Rs-Ri)*IRF
```

where Vr and Vs are the measured voltages viewing space and reference respectively, Rr is derived from the measured temperature of the reference surface, and Rs is the integrated radiance of space ( $\sim 0 \mathrm{~W} \mathrm{~cm}-2$ str- 1 ). These equations can be solved for the two unknown values, Ri and IRF, giving:

$$
\begin{aligned}
& \mathrm{Ri}=(\mathrm{Vs} * \mathrm{Rr}-\mathrm{Vr} * \mathrm{Rs}) /(\mathrm{Vs}-\mathrm{Vr}) \\
& \mathrm{IRF}=\mathrm{Vr} /(\mathrm{Rr}-\mathrm{Ri}) \\
& \mathrm{Or}: \\
& \mathrm{IRF}=\mathrm{Vs} /(\mathrm{Rs}-\mathrm{Ri})
\end{aligned}
$$

These computed values then are used to compute the radiance of the planet:

$$
R p=(V p / I R F)+R i
$$

### 4.2 THERMAL BOLOMETER ALGORITHM VERSION V001.A

The simultaneous determination of IRF and Ri requires space (S) and reference surface ( R ) observations spaced closely in time. Typically these are acquired as consecutive or interleaved observations that are termed "SR-pairs". The IRF is assumed to vary slowly, whereas Ri can vary throughout the orbit. Thus, the SRpairs are only acquired several times per orbit to determine IRF, whereas Space observations are acquired approximately every 3-5 minutes to determine Ri.

Prior to calibration, the weighted integrated radiance as a function of scene temperature is computed by convolving the instrument relative spectral response with the blackbody radiance at each wavenumber from 0 through 2500 with a step of 2 wavenumbers. The relative spectral response of the TES thermal bolometer was determined pre-launch and is given in the TES Calibration Report. The integrated radiance is computed for temperature values from 60K through 400K with a step of 0.01 degrees. A look-up table consisting of two columns: temperature and weighted integrated radiance, is stored in a separate file, and is used to convert brightness temperature (TB) to radiance and radiance to TB.

The following sequence of operations is carried out for spectral calibration:

1) Read the data associated with all the observations under consideration.
2) Find all of the single and double scan SR-pairs and Space observations (S) in the given set of observations.
3) At each SR-pair, compute the temperature of the instrument (Ti) and IRF. For each detector:
a) Average the voltage of all the Space observations having the same scan length. This is Vs.
b) Average the voltage of all the reference observations having the same scan length. This is Vr. Average the reference surface thermistor temperatures (aux_temp[1-3]) to find the average temperature of the
reference surface for this SR-pair. This is Tr .
c) Compute the radiance of the reference surface $(\mathrm{Rr})$ at temperature Tr using the TB-to-radiance look-up table.
d) Assume the radiance of space (Rs) to be equal to zero.
e) Compute IRF using the equation:

$$
\mathrm{IRF}=\mathrm{Vs} /(\mathrm{Rs}-\mathrm{Ri})
$$

f) Compute the integrated radiance of the instrument by substituting the calculated values in the equation:

$$
\mathrm{Ri}=(\mathrm{Vs} * \mathrm{Rr}-\mathrm{Vr} * \mathrm{Rs}) /(\mathrm{Vs}-\mathrm{Vr})
$$

g) Compute the instrument brightness temperature of the instrument ( Ti ) from the radiance Ri using the radiance-to-brightness temperature look-up table.
4)Store the computed values of IRF and Ti into one packet, tag it as an SRpair with its starting sclk_time and pool it among other similar packets for SRpairs and Space observations in ascending order of their sclk_time. This pool is called the IRF-pool.
5) Replicate the first SR-pair as an additional SR-pair in the beginning of the given set of observations.
6) Replicate the last SR-pair as an additional SR-pair at the end of the given set of observations.
7) At each set of Space observations, compute Vs. Vs is used to compute the radiance of the instrument ( Ri ) in the planet caibration using the equation:

$$
\mathrm{Ri}=\mathrm{Rs}-(\mathrm{Vs} / \mathrm{IRF})
$$

For each detector:
a) Average the voltage of all the Space observations in a given set of consecutive observations having the same scan length. This is Vs.
b) Store the value of Vs for this Space observation into one packet, tag this packet as an $S$ with its starting sclk_time, and pool it in ascending order of its sclk_time.
8) At each planet observation compute Rp. For each detector:
a) Interpolate linearly over sclk_time between the IRF values at the two bounding SR or S observations to determine the IRF at this planet observation.
b) Interpolate linearly over sclk_time between the Vs values at the two bounding SR or S observations to compute Vs at this planet observation. c) Compute Rp from IRF and Vs using:

$$
R p=(\mathrm{Vp} / \mathrm{IRF})+\mathrm{Ri} .
$$

Replacing Ri in this equation with $\mathrm{Ri}=\mathrm{Rs}-(\mathrm{Vs} / \operatorname{IRF})$ gives:

$$
\mathrm{Rp}=\mathrm{Rs}+(\mathrm{Vp}-\mathrm{Vs}) / \mathrm{IRF}
$$

d) Convert Rp to the brightness temperature of the planet ( Tp ) using the radiance-to-brightness temperature look-up table.
9) Write the brightness temperature to the database.

### 5.0 SURFACE TEMPERATURE DETERMINATION

A simple algorithm was performed on each TES spectrum in order to estimate the effective surface kinetic temperature using the TES spectrometer data. The primary use of this temperature is for emissivity determination where only a first-order estimate of the surface temperature is required. No attempt is made to model mixtures of surface materials at different kinetic temperatures, nor to remove atmospheric effects.

This algorithm is based on the desire to use the entire spectrum to locate the region with the highest emissivity, where the brightness temperature will provide the closest approximation to the surface kinetic temperature. In practice, both in laboratory measurements and in TES data, the short-wavelength region ( $<8$ microns) often has the highest emissivity. Unfortunately, at low temperatures (<~225 K), the short-wavelength region of the spectrum has significant noise and measurements in this spectral region are unreliable. Thus, it is necessary to have a flexible algorithm that uses the best available data to estimate surface temperature.

1) Convert the calibrated radiance to brightness temperature at each wavenumber assuming that: a) the emissivity is unity (temp. = TB); and b) the emissivity is 0.97 and dividing the calibrated radiance by this value before determining the brightness temperature (temp. $=$ TB'). Filter the brightness temperatures using a unity-weight filter 7 samples wide to reduce noise effects.
2) Find the maximum brightness temperature over the sample ranges from:
a) 300 to $1350 \mathrm{~cm}-1$, excluding the region from $500 \mathrm{~cm}-1$ to $800 \mathrm{~cm}-1$ where atmospheric CO 2 has strong absorptions. This range was selected to include both the long and short wavelength portions of the spectrum, and to include the wavenumber typically with the highest brightness temperature ( $\sim 1300 \mathrm{~cm}-1$ ) as determined by both the Mariner 9 IRIS and the preliminary TES data.
b) 300 to $500 \mathrm{~cm}-1$ only. This range covers only the long wavelength portion of the spectrum.
3) If TB is greater than or equal to $\mathrm{T} 2(225 \mathrm{~K})$, set Tsurface to TB; If TB' is less than or equal to T1 $(215 \mathrm{~K})$ set Tsurface to TB'. Otherwise, provide a smooth transition between these to cases by setting Tsurface to weighted
average of TB and TB'. Weighting is determined by:

$$
\begin{aligned}
& \text { Weight1 }=1-((\mathrm{T} 2-\mathrm{TB}) /(\mathrm{T} 2-\mathrm{T} 1)) \\
& \text { Weight2 }=1-\left(\left(\mathrm{TB}^{\prime}-\mathrm{T} 1\right) /(\mathrm{T} 2-\mathrm{T} 1)\right) \\
& \text { If Weight1 or Weight2 < } 0 \text {, then they are set to } 0 .
\end{aligned}
$$

4) Finally:

$$
\text { Tsurface }=((\text { TB*Weight } 1)+(\text { TB'*Weight } 2)) /(\text { Weight } 1+\text { Weight } 2)
$$

### 6.0 DATA QUALITY/ANOMALIES

### 6.1 SPECTRAL RINGING

The TES spectra occasionally exhibit a high frequency "ringing" in which the amplitude of the spectrum oscillates from one spectral point to the next. This ringing has been found to occur when there is a large change in scene temperature from one observation to the next. These large temperature variations frequently occur during observations of Mars acquired at large distances where there is a significant change in position on the planet between successive observations. The TES analog electronics are designed to keep the DC (base) level of interferogram centered at zero volts. However, when the radiance of the scene changes, the base level of the interferogram changes and there is a finite time required for the electronics to compensate. The frequency of the TES interferogram information band is $10-100 \mathrm{~Hz}$, so the electronics are designed to pass all information within this band to avoid filtering out signal information. Therefore, the base correction electronics are designed to have a time constant $>0.1$ seconds to avoid altering the true interferogram spectral information. There are only 0.2 seconds between the end of one interferogram and the start of the next, so if the scene changes temperature rapidly between observations, then the electronics will not have sufficient time to fully recenter the base level of the interferogram before the start of the next scan. In this case the interferogram will still be settling (or rising) toward the base level during the first $\sim 0.1-0.2$ seconds of the interferogram scan. This settling results in a discontinuity at the beginning of the scan due to the fact that the first point is significantly greater (or less) than zero. Because this spike occurs at the beginning of the scan, it always produces a sine wave at the highest possible frequency in the transformed spectrum. As a result, a sine wave with a point-to-point variation is superimposed on the data. The interferogram base level can be increasing or decreasing if the temperature of the current scan is higher or lower respectively than the previous scan. Therefore, the phase of the superimposed sine wave can vary by 180 degrees.

An algorithm has been developed to artificially remove the spectral ringing by transforming the spectral data back to frequency and removing the end points of the interferogram. However, this approximation lowers the spectral resolution of the data, and has not been applied to the calibrated spectra on the TES CDROMS in this release. A more sophisticated approach is being developed using the measured time constant of the TES electronics to model the settling of the interferogram toward the base level. This correction will be applied in later releases of the TES data.

Up to $80 \%$ of the low resolution data acquired away from periapsis can show significant ringing. Even data that do not exhibit a visible ringing can have higher than expected power in the highest frequency, suggesting that some "ringing" is present. The ringing effect should be significantly reduced when the TES is operated in the planned mode during mapping.

### 6.2 SPECTROMETER NON-ZERO BACKGROUND CALIBRATED RADIANCE

In-flight observations indicate that a small, systematic calibration offset with a magnitude of $\sim 1 \times 10-7 \mathrm{~W} \mathrm{~cm}-2$ str- $1 \mathrm{~cm}-1$ is present in the TES data presented on this CD-ROM. This error is primarily due to slight variations in the instrument background energy between observations taken of space at an angle of $-90^{\circ}$ aft ( 0 degrees $=$ nadir) for calibration and those viewing the planet at angle other than -90 degrees. This error is not significant for surface observations at temperatures above $\sim 240 \mathrm{~K}$. However, for observations of the polar caps and the atmosphere above the limbs, where the radiance is low, this error can be significant.

Observations have been collected to characterize the variation in the calibration offset with pointing mirror angle and instrument temperatures. Models are being developed to account for this effect. This correction will be applied in later releases of the TES data.

### 7.0 REFERENCES

Christensen, P.R., D.L. Anderson, J. S.C. Chase, R.N. Clark, H.H. Kieffer, M.C. Malin, J.C. Pearl, J. Carpenter, N. Bandiera, F.G. Brown, and S. Silverman, Thermal emission spectrometer experiment: Mars Observer mission, J. Geophys. Res., 97, 7719-7734, 1992.

## APPENDIX

Table A1. Spectrometer Wavenumber Position - Double Scan


|  | 40 | 355.81 | 355.59 | 355.81 | 355.09 | 355.23 | 355.09 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 21 | 41 | 361.14 | 360.91 | 361.14 | 360.39 | 360.55 | 360.39 |
|  | 42 | 366.44 | 366.20 | 366.44 | 365.68 | 365.84 | 365.68 |
| 22 | 43 | 371.73 | 371.52 | 371.73 | 370.98 | 371.16 | 370.98 |
|  | 44 | 377.06 | 376.81 | 377.06 | 376.28 | 376.45 | 376.28 |
| 23 | 45 | 382.36 | 382.13 | 382.36 | 381.58 | 381.74 | 381.58 |
|  | 46 | 387.68 | 387.45 | 387.68 | 386.87 | 387.06 | 386.87 |
| 24 | 47 | 392.98 | 392.74 | 392.98 | 392.17 | 392.35 | 392.17 |
|  | 48 | 398.28 | 398.06 | 398.28 | 397.47 | 397.67 | 397.47 |
| 25 | 49 | 403.60 | 403.35 | 403.60 | 402.76 | 402.96 | 402.76 |
|  | 50 | 408.90 | 408.67 | 408.90 | 408.09 | 408.25 | 408.09 |
| 26 | 51 | 414.23 | 413.96 | 414.23 | 413.39 | 413.57 | 413.39 |
|  | 52 | 419.52 | 419.28 | 419.52 | 418.68 | 418.86 | 418.68 |
| 27 | 53 | 424.85 | 424.60 | 424.85 | 423.98 | 424.18 | 423.98 |
|  | 54 | 430.15 | 429.89 | 430.15 | 429.28 | 429.47 | 429.28 |
| 28 | 55 | 435.45 | 435.21 | 435.45 | 434.57 | 434.76 | 434.57 |
|  | 56 | 440.77 | 440.50 | 440.77 | 439.87 | 440.08 | 439.87 |
| 29 | 57 | 446.07 | 445.82 | 446.07 | 445.17 | 445.37 | 445.17 |
|  | 58 | 451.40 | 451.14 | 451.40 | 450.47 | 450.69 | 450.47 |
| 30 | 59 | 456.69 | 456.43 | 456.69 | 455.76 | 455.98 | 455.76 |
|  | 60 | 462.02 | 461.75 | 462.02 | 461.06 | 461.27 | 461.06 |
| 31 | 61 | 467.32 | 467.05 | 467.32 | 466.36 | 466.59 | 466.36 |
|  | 62 | 472.64 | 472.37 | 472.64 | 471.65 | 471.89 | 471.65 |
| 32 | 63 | 477.94 | 477.66 | 477.94 | 476.95 | 477.21 | 476.95 |
|  | 64 | 483.24 | 482.98 | 483.24 | 482.25 | 482.50 | 482.25 |
| 33 | 65 | 488.56 | 488.30 | 488.56 | 487.54 | 487.79 | 487.54 |
|  | 66 | 493.86 | 493.59 | 493.86 | 492.84 | 493.11 | 492.84 |
| 34 | 67 | 499.19 | 498.91 | 499.19 | 498.17 | 498.40 | 498.17 |
|  | 68 | 504.49 | 504.20 | 504.49 | 503.47 | 503.69 | 503.47 |
| 35 | 69 | 509.81 | 509.52 | 509.81 | 508.76 | 509.01 | 508.76 |
|  | 70 | 515.11 | 514.84 | 515.11 | 514.06 | 514.30 | 514.06 |
| 36 | 71 | 520.41 | 520.13 | 520.41 | 519.36 | 519.62 | 519.36 |
|  | 72 | 525.73 | 525.45 | 525.73 | 524.65 | 524.91 | 524.65 |
| 37 | 73 | 531.03 | 530.74 | 531.03 | 529.95 | 530.20 | 529.95 |
|  | 74 | 536.36 | 536.06 | 536.36 | 535.25 | 535.52 | 535.25 |
| 38 | 75 | 541.65 | 541.35 | 541.65 | 540.54 | 540.81 | 540.54 |
|  | 76 | 546.98 | 546.67 | 546.98 | 545.84 | 546.13 | 545.84 |
| 39 | 77 | 552.28 | 551.99 | 552.28 | 551.14 | 551.42 | 551.14 |
|  | 78 | 557.61 | 557.28 | 557.61 | 556.43 | 556.71 | 556.43 |
| 40 | 79 | 562.90 | 562.60 | 562.90 | 561.73 | 562.03 | 561.73 |
|  | 80 | 568.23 | 567.89 | 568.23 | 567.03 | 567.32 | 567.03 |
| 41 | 81 | 573.53 | 573.21 | 573.53 | 572.36 | 572.64 | 572.36 |
|  | 82 | 578.85 | 578.50 | 578.85 | 577.65 | 577.93 | 577.65 |
| 42 | 83 | 584.15 | 583.82 | 584.15 | 582.95 | 583.22 | 582.95 |
|  | 84 | 589.48 | 589.15 | 589.48 | 588.25 | 588.55 | 588.25 |
| 43 | 85 | 594.77 | 594.44 | 594.77 | 593.54 | 593.84 | 593.54 |
|  | 86 | 600.07 | 599.76 | 600.07 | 598.84 | 599.16 | 598.84 |


| 44 | 87 | 605.40 | 605.05 | 605.40 | 604.14 | 604.45 | 604.14 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 88 | 610.69 | 610.37 | 610.69 | 609.43 | 609.74 | 609.43 |
| 45 | 89 | 616.02 | 615.66 | 616.02 | 614.73 | 615.06 | 614.73 |
|  | 90 | 621.32 | 620.98 | 621.32 | 620.03 | 620.35 | 620.03 |
| 46 | 91 | 626.65 | 626.30 | 626.65 | 625.32 | 625.67 | 625.32 |
|  | 92 | 631.94 | 631.59 | 631.94 | 630.65 | 630.96 | 630.65 |
| 47 | 93 | 637.27 | 636.91 | 637.27 | 635.95 | 636.25 | 635.95 |
|  | 94 | 642.57 | 642.20 | 642.57 | 641.25 | 641.57 | 641.25 |
| 48 | 95 | 647.86 | 647.52 | 647.86 | 646.54 | 646.86 | 646.54 |
|  | 96 | 653.19 | 652.81 | 653.19 | 651.84 | 652.18 | 651.84 |
| 49 | 97 | 658.49 | 658.13 | 658.49 | 657.14 | 657.47 | 657.14 |
|  | 98 | 663.81 | 663.42 | 663.81 | 662.43 | 662.76 | 662.43 |
| 50 | 99 | 669.11 | 668.74 | 669.11 | 667.73 | 668.08 | 667.73 |
|  | 100 | 674.44 | 674.06 | 674.44 | 673.03 | 673.37 | 673.03 |
| 51 | 101 | 679.74 | 679.35 | 679.74 | 678.32 | 678.69 | 678.32 |
|  | 102 | 685.06 | 684.67 | 685.06 | 683.62 | 683.98 | 683.62 |
| 52 | 103 | 690.36 | 689.96 | 690.36 | 688.95 | 689.27 | 688.95 |
|  | 104 | 695.66 | 695.28 | 695.66 | 694.25 | 694.59 | 694.25 |
| 53 | 105 | 700.98 | 700.57 | 700.98 | 699.54 | 699.88 | 699.54 |
|  | 106 | 706.28 | 705.90 | 706.28 | 704.84 | 705.20 | 704.84 |
| 54 | 107 | 711.61 | 711.22 | 711.61 | 710.14 | 710.50 | 710.14 |
|  | 108 | 716.90 | 716.51 | 716.90 | 715.43 | 715.79 | 715.43 |
| 55 | 109 | 722.23 | 721.83 | 722.23 | 720.73 | 721.11 | 720.73 |
|  | 110 | 727.53 | 727.12 | 727.53 | 726.03 | 726.40 | 726.03 |
| 56 | 111 | 732.82 | 732.44 | 732.82 | 731.32 | 731.72 | 731.32 |
|  | 112 | 738.15 | 737.76 | 738.15 | 736.62 | 737.0 | 736.62 |
| 57 | 113 | 743.45 | 743.05 | 743.45 | 741.95 | 742.30 | 741.95 |
|  | 114 | 748.78 | 748.37 | 748.78 | 747.24 | 747.62 | 747.24 |
| 58 | 115 | 754.07 | 753.66 | 754.07 | 752.54 | 752.91 | 752.54 |
|  | 116 | 759.37 | 758.98 | 759.37 | 757.84 | 758.23 | 757.84 |
| 59 | 117 | 764.70 | 764.27 | 764.70 | 763.14 | 763.52 | 763.14 |
|  | 118 | 769.99 | 769.59 | 769.99 | 768.43 | 768.81 | 768.43 |
| 60 | 119 | 775.32 | 774.91 | 775.32 | 773.73 | 774.13 | 773.73 |
|  | 120 | 780.62 | 780.20 | 780.62 | 779.03 | 779.42 | 779.03 |
| 61 | 121 | 785.94 | 785.52 | 785.94 | 784.32 | 784.74 | 784.32 |
|  | 122 | 791.24 | 790.81 | 791.24 | 789.62 | 790.03 | 789.62 |
| 62 | 123 | 796.54 | 796.13 | 796.54 | 794.92 | 795.32 | 794.92 |
|  | 124 | 801.86 | 801.42 | 801.86 | 800.24 | 800.64 | 800.24 |
| 63 | 125 | 807.16 | 806.74 | 807.16 | 805.54 | 805.93 | 805.54 |
|  | 126 | 812.49 | 812.06 | 812.49 | 810.84 | 811.25 | 810.84 |
| 64 | 127 | 817.79 | 817.35 | 817.79 | 816.13 | 816.54 | 816.13 |
|  | 128 | 823.11 | 822.68 | 823.11 | 821.43 | 821.83 | 821.43 |
| 65 | 129 | 828.41 | 827.97 | 828.41 | 826.73 | 827.16 | 826.73 |
|  | 130 | 833.71 | 833.29 | 833.71 | 832.03 | 832.45 | 832.03 |
| 66 | 131 | 839.03 | 838.58 | 839.03 | 837.32 | 837.77 | 837.32 |
|  | 132 | 844.33 | 843.90 | 844.33 | 842.62 | 843.06 | 842.62 |
| 67 | 133 | 849.66 | 849.22 | 849.66 | 847.92 | 848.35 | 847.92 |


|  | 134 | 854.95 | 854.51 | 854.95 | 853.21 | 853.678 | 853.21 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 68 | 135 | 860.25 | 859.83 | 860.25 | 858.51 | 858.968 | 858.51 |
|  | 136 | 865.58 | 865.12 | 865.58 | 863.81 | 864.28 | 863.81 |
| 69 | 137 | 870.88 | 870.44 | 870.88 | 869.108 | 869.57 | 869.10 |
|  | 138 | 876.20 | 875.73 | 876.20 | 874.43 | 874.86 | 874.43 |
| 70 | 139 | 881.50 | 881.05 | 881.50 | 879.73 | 880.18 | 879.73 |
|  | 140 | 886.80 | 886.37 | 886.80 | 885.03 | 885.47 | 885.03 |
| 71 | 141 | 892.12 | 891.66 | 892.12 | 890.328 | 890.798 | 890.32 |
|  | 142 | 897.42 | 896.98 | 897.42 | 895.628 | 896.08 | 895.62 |
| 72 | 143 | 902.75 | 902.27 | 902.75 | 900.929 | 901.379 | 900.92 |
|  | 144 | 908.04 | 907.59 | 908.04 | 906.21 | 906.69 | 906.21 |
| 73 | 145 | 913.37 | 912.88 | 913.37 | 911.51 | 911.98 | 911.51 |
|  | 146 | 918.67 | 918.20 | 918.67 | 916.81 | 917.27 | 916.81 |
| 74 | 147 | 923.96 | 923.49 | 923.96 | 922.10 | 922.59 | 922.10 |
|  | 148 | 929.29 | 928.81 | 929.29 | 927.40 | 927.88 | 927.40 |
| 75 | 149 | 934.59 | 934.13 | 934.59 | 932.70 | 933.20 | 932.70 |
|  | 150 | 939.92 | 939.43 | 939.92 | 937.99 | 938.49 | 937.99 |
| 76 | 151 | 945.21 | 944.75 | 945.21 | 943.32 | 943.79 | 943.32 |
|  | 152 | 950.51 | 950.04 | 950.51 | 948.629 | 949.119 | 948.62 |
| 77 | 153 | 955.84 | 955.36 | 955.84 | 953.929 | 954.40 | 953.92 |
|  | 154 | 961.13 | 960.65 | 961.13 | 959.21 | 959.72 | 959.21 |
| 78 | 155 | 966.46 | 965.97 | 966.46 | 964.51 | 965.01 | 964.51 |
|  | 156 | 971.76 | 971.26 | 971.76 | 969.81 | 970.30 | 969.81 |
| 79 | 157 | 977.08 | 976.58 | 977.08 | 975.10 | 975.62 | 975.10 |
|  | 158 | 982.38 | 981.90 | 982.38 | 980.40 | 980.91 | 980.40 |
| 80 | 159 | 987.68 | 987.19 | 987.68 | 985.70 | 986.23 | 985.70 |
|  | 160 | 993.01 | 992.51 | 993.01 | 990.99 | 991.52 | 990.99 |
| 81 | 161 | 998.30 | 997.80 | 998.30 | 996.29 | 996.81 | 996.29 |
|  | 162 | 1003.63 | 1003.12 | 1003.63 | 1001.59 | 591002.13 | 31001.59 |
| 82 | 163 | 1008.93 | 1008.41 | 1008.93 | 1006.88 | 1007.42 | 1006.88 |
|  | 164 | 1014.22 | 1013.73 | 1014.22 | 21012.18 | 81012.74 | ( 741012.18 |
| 83 | 165 | 1019.55 | 1019.05 | 1019.55 | 51017.48 | 48 1018.03 | 81017.48 |
|  | 166 | 1024.85 | 1024.34 | 1024.85 | 1022.78 | 7 1023.32 | 32 1022.78 |
| 84 | 167 | 1030.17 | 1029.66 | 1030.17 | 71028.10 | O 1028.6 | 1028.10 |
|  | 168 | 1035.47 | 1034.95 | 1035.47 | 1033.40 | 40 1033.93 | 93 1033.40 |
| 85 | 169 | 1040.77 | 1040.27 | 1040.77 | 1038.70 | 70 1039.25 | 251038.70 |
|  | 170 | 1046.09 | 1045.56 | 1046.09 | 1043.99 | 991044.5 | 541043.99 |
| 86 | 171 | 1051.39 | 1050.88 | 1051.39 | 1049.29 | 291049.83 | 831049.29 |
|  | 172 | 1056.72 | 1056.17 | 1056.72 | 21054.59 | 1055.15 | 51054.59 |
| 87 | 173 | 1062.01 | 1061.49 | 1062.01 | 11059.88 | 881060.4 | . 1059.88 |
|  | 174 | 1067.34 | 1066.82 | 1067.34 | 41065.18 | 81065.76 | 76 1065.18 |
| 88 | 175 | 1072.64 | 1072.11 | 1072.64 | 1070.48 | 1071.06 | 1070.48 |
|  | 176 | 1077.94 | 1077.43 | 1077.94 | 1075.77 | 71076.35 | 35 1075.77 |
| 89 | 177 | 1083.26 | 1082.72 | 1083.26 | 61081.07 | 1081.6 | 67 1081.07 |
|  | 178 | 1088.56 | 1088.04 | 1088.56 | 61086.37 | 1086.96 | 961086.37 |
| 90 | 179 | 1093.89 | 1093.33 | 1093.89 | 1091.67 | 1092.25 | 25 1091.67 |
|  | 180 | 1099.18 | 1098.65 | 1099.18 | 81096.99 | 1097.5 | 1096.99 |


| 91 | 181 | 1104.48 | 1103.97 | 1104.48 | 1102 | 1102.86 | 1102.29 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 182 | 1109.81 | 1109.26 | 1109.81 | 1107.59 | 1108.18 | 1107.59 |
| 92 | 183 | 1115.10 | 1114.58 | 1115.10 | 1112.88 | 1113.47 | 1112.88 |
|  | 4 | 1120.40 | 1119.87 | 1120.40 | 1118.18 | 1118.76 | 8. |
| 93 | 185 | 1125.73 | 1125.19 | 1125.73 | 1123.48 | 1124.08 | 1123.48 |
|  | 186 | 1131.03 | 1130.48 | 1131.03 | 1128.77 | 7 | 1128.77 |
| 94 | 187 | 1136.35 | 1135.80 | 1136.35 | 1134.07 | 1134.69 | 1134.07 |
|  | 188 | 1141.65 | 1141.12 | 1141.65 | 1139.37 | 1139.98 | 1139.37 |
| 95 | 189 | 1146.95 | 1146.41 | 1146.95 | 1144.66 | 1145.27 | 1144.66 |
|  | 190 | 1152.27 | 1151.73 | 1152.27 | 1149.96 | 1150.59 | 1149.96 |
| 96 | 191 | 1157.57 | 1157.02 | 1157.57 | 1155.26 | 1155.88 | 1155.26 |
|  | 192 | 1162.90 | 1162.34 | 1162.90 | 1160.56 | 1161.20 | 1160.56 |
| 97 | 193 | 1168.19 | 1167.63 | 1168.19 | 1165.85 | 1166.49 | 165.85 |
|  | 194 | 1173.49 | 1172.95 | 1173.49 | 1171.15 | 1171.78 |  |
| 98 | 195 | 1178.82 | 1178.24 | 1178.82 | 1176.48 | 1177.10 | 176.48 |
|  | 196 | 1184.11 | 1183.57 | 1184.11 | 1181.77 | 1182.39 | 77 |
| 99 | 197 | 1189.44 | 1188.89 | 1189.44 | 1187.07 | 1187.72 | 1187.07 |
|  | 198 | 1194.74 | 1194.18 | 1194.74 | 1192.37 | 1193.01 | 1192.37 |
| 100 | 199 | 1200.04 | 1199.50 | 1200.04 | 1197.66 | 1198.30 | 7.66 |
|  | 200 | 1205.36 | 1204.79 | 1205.36 | 1202.96 | 1203.62 | 202.96 |
| 101 | 201 | 1210.66 | 1210.11 | 1210.66 | 1208.26 | 1208.91 | 1208.26 |
|  | 202 | 1215.99 | 1215.40 | 1215.99 | 1213.56 | 1214.23 | 213.56 |
| 102 | 203 | 1221.28 | 1220.72 | 1221.28 | 1218.85 | 1219.52 | 218.8 |
|  | 204 | 1226.58 | 1226.01 | 1226.58 | 1224.15 | 1224.81 |  |
| 103 | 205 | 1231.91 | 1231.33 | 1231.91 | 1229.45 | 1230.13 | 1229.4 |
|  | 206 | 1237.20 | 1236.65 | 1237.20 | 1234.74 | 1235.42 |  |
| 104 | 207 | 1242.53 | 1241.94 | 1242.53 | 1240.04 | 1240.74 | 1240.0 |
|  | 208 | 1247.83 | 1247.26 | 1247.83 | 1245.34 | 1246.03 | 1245.3 |
| 105 | 209 | 1253.13 | 1252.55 | 1253.13 | 1250.66 | 1251.32 |  |
|  | 210 | 1258.45 | 1257.87 | 1258.45 | 1255.96 | 1256.64 |  |
| 106 | 211 | 1263.75 | 1263.16 | 1263.75 | 1261.26 | 1261.93 | 1261.26 |
|  | 212 | 1269.08 | 1268.48 | 1269.08 | 1266.55 | 1267.25 | 126 |
| 107 | 213 | 1274.37 | 1273.77 | 1274.37 | 1271.85 | 1272.54 | 1271.8 |
|  | 214 | 1279.67 | 1279.09 | 1279.67 | 1277.15 | 1277.83 |  |
| 108 | 215 | 1285.00 | 1284.38 | 1285.00 | 1282.45 | 1283.15 | 1282.4 |
|  | 216 | 1290.29 | 1289.70 | 1290.29 | 1287.74 | 1288.44 |  |
| 109 | 217 | 1295.62 | 1295.02 | 1295.62 | 1293.0 | 1293.76 |  |
|  | 218 | 1300.92 | 1300.32 | 1300.92 | 1298.34 | 1299.05 |  |
| 110 | 219 | 1306.21 | 1305.64 | 1306.21 | 1303.63 | 1304.34 | 1303.63 |
|  | 220 | 1311.54 | 1310.93 | 1311.54 | 1308.93 | 1309.67 | 308.9 |
| 111 | 221 | 1316.84 | 1316.25 | 1316.84 | 1314.23 | 1314.96 | 314. |
|  | 222 | 1322.17 | 1321.54 | 1322.17 | 1319.52 | 1320.25 | 1319.52 |
| 112 | 223 | 1327.46 | 1326.86 | 1327.46 | 1324.82 | 1325.57 | 1324.8 |
|  | 224 | 1332.76 | 1332.15 | 1332.76 | 1330.12 | 1330.86 | 1330.12 |
| 113 | 225 | 1338.09 | 1337.47 | 1338.09 | 1335.41 | 1336.18 | 1335. |
|  | 226 | 1343.38 | 1342.79 | 1343.38 | 1340.71 | 1341.47 | 1340 |
| 114 | 227 | 1348.71 | 1348.08 | 1348.71 | 1346.01 | 1346.76 | 34 |


|  | 228 | 1354.01 |  | 1354.01 | 1351.31 | 1352.08 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 115 | 229 | 1359.30 | 1358.69 | 1359.30 | 1356.63 | 1357.37 | 1356.63 |
|  | 230 | 1364.63 | 1364.01 | 1364.63 | 1361.93 | 1362.69 | 1361.93 |
| 116 | 231 | 1369.93 | 1369.30 | 1369.93 | 1367.23 | 1367.98 | 1367.23 |
|  | 232 | 1375.25 | 1374.62 | 1375.25 | 1372.52 | 1373.27 | 1372.52 |
| 117 | 233 | 1380.55 | 1379.91 | 1380.55 | 1377.82 | 1378.59 | 1377.82 |
|  | 234 | 1385.88 | 1385.23 | 1385.88 | 1383.12 | 1383.88 | 1383.12 |
| 118 | 235 | 1391.18 | 1390.55 | 1391.18 | 1388.41 | 1389.17 | 1388.41 |
|  | 236 | 1396.47 | 1395.84 | 1396.47 | 1393.71 | 1394.49 | 1393.71 |
| 119 | 237 | 1401.80 | 1401.16 | 1401.80 | 1399.01 | 1399.78 | 1399.01 |
|  | 238 | 1407.10 | 1406.45 | 1407.10 | 1404.31 | 1405.10 | 1404.31 |
| 120 | 239 | 1412.42 | 1411.77 | 1412.42 | 1409.60 | 1410.39 | 1409.60 |
|  | 240 | 1417.72 | 1417.06 | 1417.72 | 1414.90 | 1415.68 | 1414.90 |
| 121 | 241 | 1423.05 | 1422.39 | 1423.05 | 1420.20 | 1421.00 | 1420.20 |
|  | 242 | 1428.34 | 1427.68 | 1428.34 | 1425.49 | 1426.30 | 1425.49 |
| 122 | 243 | 1433.64 | 1433.00 | 1433.64 | 1430.79 | 1431.62 | 1430.79 |
|  | 244 | 1438.97 | 1438.32 | 1438.97 | 1436.09 | 1436.91 | 1436.09 |
| 123 | 245 | 1444.27 | 1443.61 | 1444.27 | 1441.38 | 1442.20 | 1441.38 |
|  | 246 | 1449.59 | 1448.93 | 1449.59 | 1446.68 | 1447.52 | 1446.68 |
| 124 | 247 | 1454.89 | 1454.22 | 1454.89 | 1451.98 | 1452.81 | 1451.98 |
|  | 248 | 1460.22 | 1459.54 | 1460.22 | 1457.27 | 1458.13 | 1457.27 |
| 125 | 249 | 1465.51 | 1464.83 | 1465.51 | 1462.57 | 1463.42 | 1462.57 |
|  | 250 | 1470.81 | 1470.15 | 1470.81 | 1467.87 | 1468.71 | 1467.87 |
| 126 | 251 | 1476.14 | 1475.44 | 1476.14 | 1473.17 | 1474.03 | 1473.17 |
|  | 252 | 1481.43 | 1480.76 | 1481.43 | 1478.49 | 1479.32 | 1478.49 |
| 127 | 253 | 1486.76 | 1486.05 | 1486.76 | 1483.79 | 1484.61 | 1483.79 |
|  | 254 | 1492.06 | 1491.37 | 1492.06 | 1489.09 | 1489.93 | 1489.09 |
| 128 | 255 | 1497.38 | 1496.69 | 1497.38 | 1494.38 | 1495.22 | 1494.38 |
|  | 256 | 1502.68 | 1501.98 | 1502.68 | 1499.68 | 1500.54 | 1499.68 |
| 129 | 257 | 1508.01 | 1507.30 | 1508.01 | 1504.98 | 1505.83 | 1504.98 |
|  | 258 | 1513.31 | 1512.59 | 1513.31 | 1510.27 | 1511.12 | 1510.27 |
| 130 | 259 | 1518.63 | 1517.91 | 1518.63 | 1515.57 | 1516.44 | 1515.57 |
|  | 260 | 1523.93 | 1523.20 | 1523.93 | 1520.87 | 1521.73 | 1520.87 |
| 131 | 261 | 1529.26 | 1528.52 | 1529.26 | 1526.16 | 1527.05 | 1526.16 |
|  | 262 | 1534.55 | 1533.81 | 1534.55 | 1531.46 | 1532.34 | 1531.46 |
| 132 | 263 | 1539.88 | 1539.14 | 1539.88 | 1536.76 | 1537.63 | 1536.76 |
|  | 264 | 1545.18 | 1544.43 | 1545.18 | 1542.06 | 1542.96 | 1542.06 |
| 133 | 265 | 1550.50 | 1549.75 | 1550.50 | 1547.35 | 1548.25 | 1547.35 |
|  | 266 | 1555.80 | 1555.07 | 1555.80 | 1552.65 | 1553.57 | 1552.65 |
| 134 | 267 | 1561.13 | 1560.36 | 1561.13 | 1557.95 | 1558.86 | 1557.95 |
|  | 268 | 1566.43 | 1565.68 | 1566.43 | 1563.24 | 1564.15 | 1563.24 |
| 135 | 269 | 1571.75 | 1570.97 | 1571.75 | 1568.54 | 1569.47 | 1568.54 |
|  | 270 | 1577.05 | 1576.29 | 1577.05 | 1573.84 | 1574.76 | 1573.84 |
| 136 | 271 | 1582.38 | 1581.58 | 1582.38 | 1579.13 | 1580.08 | 1579.13 |
|  | 272 | 1587.67 | 1586.90 | 1587.67 | 1584.43 | 1585.37 | 1584.43 |
| 137 | 273 | 1593.00 | 1592.19 | 1593.00 | 1589.73 | 1590.66 | 1589.73 |
|  | 274 | 1598.30 | 1597.51 | 1598.30 | 1595.03 | 1595.98 | 1595.03 |


| 138 | 275 | 1603.62 | 1602.80 | 1603.62 | 1600.32 | 1601.27 | 1600.32 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
|  | 276 | 1608.92 | 1608.12 | 1608.92 | 1605.62 | 1606.59 | 1605.62 |
| 139 | 277 | 1614.25 | 1613.41 | 1614.25 | 1610.92 | 1611.88 | 1610.92 |
|  | 278 | 1619.54 | 1618.73 | 1619.54 | 1616.21 | 1617.17 | 1616.21 |
| 140 | 279 | 1624.87 | 1624.05 | 1624.87 | 1621.51 | 1622.49 | 1621.51 |
|  | 280 | 1630.17 | 1629.34 | 1630.17 | 1626.81 | 1627.78 | 1626.81 |
| 141 | 281 | 1635.50 | 1634.66 | 1635.50 | 1632.10 | 1633.07 | 1632.10 |
|  | 282 | 1640.82 | 1639.95 | 1640.82 | 1637.40 | 1638.39 | 1637.40 |
| 142 | 283 | 1646.12 | 1645.27 | 1646.12 | 1642.70 | 1643.68 | 1642.70 |
|  | 284 | 1651.45 | 1650.56 | 1651.45 | 1647.99 | 1649.00 | 1647.99 |
| 143 | 285 | 1656.74 | 1655.89 | 1656.74 | 1653.29 | 1654.29 | 1653.29 |
|  | 286 | 1662.07 | 1661.18 | 1662.07 | 1658.59 | 1659.58 | 1658.59 |
| 144 | 287 | 1667.40 | 1666.50 | 1667.40 | 1663.89 | 1664.91 | 1663.89 |
|  | 288 | 1672.69 | 1671.79 | 1672.69 | 1669.15 | 1670.20 | 1669.15 |
| 145 | 289 | 1678.02 | 1677.11 | 1678.02 | 1674.45 | 1675.52 | 1674.45 |
|  | 290 | 1683.32 | 1682.40 | 1683.32 | 1679.75 | 1680.81 | 1679.75 |
| 146 | 291 | 1688.64 | 1687.72 | 1688.64 | 1685.04 | 1686.10 | 1685.04 |
|  | 292 | 1693.97 | 1693.04 | 1693.97 | 1690.34 | 1691.42 | 1690.34 |
| 147 | 293 | 1699.27 | 1698.33 | 1699.27 | 1695.64 | 1696.71 | 1695.64 |
|  | 294 | 1704.60 | 1703.65 | 1704.60 | 1700.93 | 1702.03 | 1700.93 |
| 148 | 295 | 1709.92 | 1708.94 | 1709.92 | 1706.23 | 1707.32 | 1706.23 |
|  | 296 | 1715.22 | 1714.26 | 1715.22 | 1711.53 | 1712.61 | 1711.53 |

Table A2. Spectrometer Line Shape. Full-width Half-maximum. Double Scan.

Det. 1 Det. 2 Det. 3 Det. 4 Det. 5 Det. 6

| 1 | 6.33 | 6.24 | 6.33 | 6.30 | 6.24 | 6.30 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 2 | 6.33 | 6.24 | 6.33 | 6.30 | 6.21 | 6.30 |
| 3 | 6.33 | 6.24 | 6.33 | 6.33 | 6.24 | 6.33 |
| 4 | 6.33 | 6.27 | 6.33 | 6.33 | 6.24 | 6.33 |
| 5 | 6.36 | 6.24 | 6.36 | 6.33 | 6.21 | 6.33 |
| 6 | 6.33 | 6.27 | 6.33 | 6.33 | 6.24 | 6.33 |
| 7 | 6.36 | 6.27 | 6.36 | 6.33 | 6.24 | 6.33 |
| 8 | 6.36 | 6.27 | 6.36 | 6.33 | 6.24 | 6.33 |
| 9 | 6.36 | 6.27 | 6.36 | 6.33 | 6.24 | 6.33 |
| 10 | 6.39 | 6.27 | 6.39 | 6.33 | 6.27 | 6.33 |
| 11 | 6.36 | 6.27 | 6.36 | 6.33 | 6.24 | 6.33 |
| 12 | 6.39 | 6.27 | 6.39 | 6.36 | 6.27 | 6.36 |
| 13 | 6.36 | 6.27 | 6.36 | 6.36 | 6.27 | 6.36 |
| 14 | 6.39 | 6.27 | 6.39 | 6.36 | 6.27 | 6.36 |
| 15 | 6.36 | 6.30 | 6.36 | 6.36 | 6.27 | 6.36 |
| 16 | 6.39 | 6.27 | 6.39 | 6.36 | 6.27 | 6.36 |
| 17 | 6.39 | 6.30 | 6.39 | 6.39 | 6.27 | 6.39 |
| 18 | 6.39 | 6.30 | 6.39 | 6.39 | 6.27 | 6.39 |


| 19 | 6.39 | 6.30 | 6.39 | 6.39 | 6.27 | 6.39 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 20 | 6.39 | 6.30 | 6.39 | 6.39 | 6.27 | 6.39 |
| 21 | 6.39 | 6.30 | 6.39 | 6.36 | 6.30 | 6.36 |
| 22 | 6.39 | 6.30 | 6.39 | 6.36 | 6.27 | 6.36 |
| 23 | 6.39 | 6.33 | 6.39 | 6.39 | 6.30 | 6.39 |
| 24 | 6.42 | 6.30 | 6.42 | 6.39 | 6.30 | 6.39 |
| 25 | 6.42 | 6.33 | 6.42 | 6.39 | 6.27 | 6.39 |
| 26 | 6.42 | 6.33 | 6.42 | 6.39 | 6.30 | 6.39 |
| 27 | 6.42 | 6.33 | 6.42 | 6.39 | 6.30 | 6.39 |
| 28 | 6.42 | 6.33 | 6.42 | 6.39 | 6.30 | 6.39 |
| 29 | 6.45 | 6.33 | 6.45 | 6.39 | 6.30 | 6.39 |
| 30 | 6.42 | 6.33 | 6.42 | 6.42 | 6.30 | 6.42 |
| 31 | 6.45 | 6.33 | 6.45 | 6.42 | 6.30 | 6.42 |
| 32 | 6.45 | 6.33 | 6.45 | 6.42 | 6.30 | 6.42 |
| 33 | 6.45 | 6.33 | 6.45 | 6.42 | 6.30 | 6.42 |
| 34 | 6.45 | 6.36 | 6.45 | 6.42 | 6.30 | 6.42 |
| 35 | 6.48 | 6.33 | 6.48 | 6.42 | 6.33 | 6.42 |
| 36 | 6.45 | 6.36 | 6.45 | 6.42 | 6.30 | 6.42 |
| 37 | 6.48 | 6.33 | 6.48 | 6.45 | 6.33 | 6.45 |
| 38 | 6.48 | 6.36 | 6.48 | 6.45 | 6.33 | 6.45 |
| 39 | 6.48 | 6.36 | 6.48 | 6.45 | 6.30 | 6.45 |
| 40 | 6.48 | 6.36 | 6.48 | 6.45 | 6.33 | 6.45 |
| 41 | 6.51 | 6.36 | 6.51 | 6.42 | 6.30 | 6.42 |
| 42 | 6.48 | 6.36 | 6.48 | 6.42 | 6.33 | 6.42 |
| 43 | 6.51 | 6.36 | 6.51 | 6.42 | 6.33 | 6.42 |
| 44 | 6.51 | 6.36 | 6.51 | 6.42 | 6.33 | 6.42 |
| 45 | 6.51 | 6.36 | 6.51 | 6.45 | 6.33 | 6.45 |
| 46 | 6.51 | 6.36 | 6.51 | 6.45 | 6.33 | 6.45 |
| 47 | 6.54 | 6.39 | 6.54 | 6.45 | 6.33 | 6.45 |
| 48 | 6.51 | 6.36 | 6.51 | 6.45 | 6.33 | 6.45 |
| 49 | 6.54 | 6.39 | 6.54 | 6.45 | 6.33 | 6.45 |
| 50 | 6.57 | 6.36 | 6.57 | 6.45 | 6.33 | 6.45 |
| 51 | 6.54 | 6.39 | 6.54 | 6.45 | 6.33 | 6.45 |
| 52 | 6.57 | 6.39 | 6.57 | 6.48 | 6.33 | 6.48 |
| 53 | 6.57 | 6.39 | 6.57 | 6.48 | 6.33 | 6.48 |
| 54 | 6.60 | 6.39 | 6.60 | 6.48 | 6.36 | 6.48 |
| 55 | 6.57 | 6.39 | 6.57 | 6.48 | 6.33 | 6.48 |
| 56 | 6.60 | 6.39 | 6.60 | 6.51 | 6.36 | 6.51 |
| 57 | 6.63 | 6.42 | 6.63 | 6.51 | 6.33 | 6.51 |
| 58 | 6.60 | 6.39 | 6.60 | 6.51 | 6.33 | 6.51 |
| 59 | 6.63 | 6.39 | 6.63 | 6.54 | 6.36 | 6.54 |
| 60 | 6.63 | 6.42 | 6.63 | 6.54 | 6.33 | 6.54 |
| 61 | 6.66 | 6.39 | 6.66 | 6.54 | 6.36 | 6.54 |
| 62 | 6.66 | 6.42 | 6.66 | 6.54 | 6.36 | 6.54 |
| 63 | 6.66 | 6.39 | 6.66 | 6.54 | 6.33 | 6.54 |
| 64 | 6.69 | 6.42 | 6.69 | 6.54 | 6.36 | 6.54 |
| 65 | 6.66 | 6.42 | 6.66 | 6.57 | 6.33 | 6.57 |
|  |  |  |  |  |  |  |


| 66 | 6.69 | 6.42 | 6.69 | 6.57 | 6.36 | 6.57 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 67 | 6.72 | 6.42 | 6.72 | 6.57 | 6.36 | 6.57 |
| 68 | 6.72 | 6.42 | 6.72 | 6.57 | 6.36 | 6.57 |
| 69 | 6.72 | 6.42 | 6.72 | 6.57 | 6.36 | 6.57 |
| 70 | 6.72 | 6.45 | 6.72 | 6.60 | 6.33 | 6.60 |
| 71 | 6.75 | 6.42 | 6.75 | 6.60 | 6.36 | 6.60 |
| 72 | 6.75 | 6.45 | 6.75 | 6.60 | 6.36 | 6.60 |
| 73 | 6.75 | 6.42 | 6.75 | 6.60 | 6.36 | 6.60 |
| 74 | 6.78 | 6.42 | 6.78 | 6.60 | 6.36 | 6.60 |
| 75 | 6.81 | 6.45 | 6.81 | 6.63 | 6.36 | 6.63 |
| 76 | 6.78 | 6.42 | 6.78 | 6.63 | 6.36 | 6.63 |
| 77 | 6.81 | 6.45 | 6.81 | 6.63 | 6.36 | 6.63 |
| 78 | 6.84 | 6.45 | 6.84 | 6.63 | 6.36 | 6.63 |
| 79 | 6.84 | 6.45 | 6.84 | 6.63 | 6.36 | 6.63 |
| 80 | 6.84 | 6.45 | 6.84 | 6.66 | 6.39 | 6.66 |
| 81 | 6.84 | 6.45 | 6.84 | 6.66 | 6.36 | 6.66 |
| 82 | 6.87 | 6.45 | 6.87 | 6.69 | 6.36 | 6.69 |
| 83 | 6.90 | 6.48 | 6.90 | 6.69 | 6.39 | 6.69 |
| 84 | 6.87 | 6.45 | 6.87 | 6.69 | 6.36 | 6.69 |
| 85 | 6.90 | 6.48 | 6.90 | 6.72 | 6.39 | 6.72 |
| 86 | 6.93 | 6.48 | 6.93 | 6.72 | 6.36 | 6.72 |
| 87 | 6.90 | 6.48 | 6.90 | 6.72 | 6.36 | 6.72 |
| 88 | 6.93 | 6.48 | 6.93 | 6.72 | 6.39 | 6.72 |
| 89 | 6.96 | 6.45 | 6.96 | 6.75 | 6.36 | 6.75 |
| 90 | 6.96 | 6.48 | 6.96 | 6.75 | 6.39 | 6.75 |
| 91 | 6.96 | 6.48 | 6.96 | 6.75 | 6.39 | 6.75 |
| 92 | 6.99 | 6.48 | 6.99 | 6.75 | 6.39 | 6.75 |
| 93 | 6.99 | 6.48 | 6.99 | 6.78 | 6.39 | 6.78 |
| 94 | 7.02 | 6.48 | 7.02 | 6.78 | 6.36 | 6.78 |
| 95 | 6.99 | 6.48 | 6.99 | 6.78 | 6.39 | 6.78 |
| 96 | 7.02 | 6.51 | 7.02 | 6.78 | 6.39 | 6.78 |
| 97 | 7.05 | 6.48 | 7.05 | 6.78 | 6.39 | 6.78 |
| 98 | 7.05 | 6.51 | 7.05 | 6.81 | 6.39 | 6.81 |
| 99 | 7.08 | 6.51 | 7.08 | 6.81 | 6.39 | 6.81 |
| 100 | 7.08 | 6.51 | 7.08 | 6.84 | 6.39 | 6.84 |
| 101 | 7.08 | 6.51 | 7.08 | 6.84 | 6.39 | 6.84 |
| 102 | 7.11 | 6.51 | 7.11 | 6.87 | 6.39 | 6.87 |
| 103 | 7.14 | 6.51 | 7.14 | 6.87 | 6.39 | 6.87 |
| 104 | 7.11 | 6.54 | 7.11 | 6.87 | 6.39 | 6.87 |
| 105 | 7.14 | 6.51 | 7.14 | 6.87 | 6.39 | 6.87 |
| 106 | 7.17 | 6.54 | 7.17 | 6.90 | 6.39 | 6.90 |
| 107 | 7.17 | 6.54 | 7.17 | 6.90 | 6.42 | 6.90 |
| 108 | 7.20 | 6.54 | 7.20 | 6.90 | 6.39 | 6.90 |
| 109 | 7.20 | 6.54 | 7.20 | 6.93 | 6.42 | 6.93 |
| 110 | 7.20 | 6.54 | 7.20 | 6.93 | 6.39 | 6.93 |
| 111 | 7.23 | 6.54 | 7.23 | 6.93 | 6.39 | 6.93 |
| 112 | 7.26 | 6.54 | 7.26 | 6.93 | 6.42 | 6.93 |
|  |  |  |  |  |  |  |


| 113 | 7.26 | 6.54 | 7.26 | 6.96 | 6.39 | 6.96 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 114 | 7.29 | 6.54 | 7.29 | 6.99 | 6.42 | 6.99 |
| 115 | 7.29 | 6.54 | 7.29 | 6.99 | 6.42 | 6.99 |
| 116 | 7.32 | 6.54 | 7.32 | 6.99 | 6.39 | 6.99 |
| 117 | 7.32 | 6.57 | 7.32 | 7.02 | 6.42 | 7.02 |
| 118 | 7.35 | 6.54 | 7.35 | 7.02 | 6.39 | 7.02 |
| 119 | 7.38 | 6.57 | 7.38 | 7.02 | 6.42 | 7.02 |
| 120 | 7.38 | 6.57 | 7.38 | 7.02 | 6.42 | 7.02 |
| 121 | 7.38 | 6.57 | 7.38 | 7.05 | 6.42 | 7.05 |
| 122 | 7.41 | 6.57 | 7.41 | 7.05 | 6.42 | 7.05 |
| 123 | 7.41 | 6.57 | 7.41 | 7.05 | 6.42 | 7.05 |
| 124 | 7.44 | 6.57 | 7.44 | 7.08 | 6.42 | 7.08 |
| 125 | 7.47 | 6.60 | 7.47 | 7.11 | 6.42 | 7.11 |
| 126 | 7.50 | 6.57 | 7.50 | 7.11 | 6.42 | 7.11 |
| 127 | 7.50 | 6.60 | 7.50 | 7.11 | 6.42 | 7.11 |
| 128 | 7.53 | 6.60 | 7.53 | 7.14 | 6.42 | 7.14 |
| 129 | 7.53 | 6.60 | 7.53 | 7.14 | 6.42 | 7.14 |
| 130 | 7.56 | 6.60 | 7.56 | 7.14 | 6.42 | 7.14 |
| 131 | 7.56 | 6.63 | 7.56 | 7.14 | 6.45 | 7.14 |
| 132 | 7.59 | 6.60 | 7.59 | 7.17 | 6.42 | 7.17 |
| 133 | 7.62 | 6.63 | 7.62 | 7.20 | 6.45 | 7.20 |
| 134 | 7.65 | 6.60 | 7.65 | 7.20 | 6.42 | 7.20 |
| 135 | 7.65 | 6.63 | 7.65 | 7.23 | 6.42 | 7.23 |
| 136 | 7.68 | 6.63 | 7.68 | 7.23 | 6.45 | 7.23 |
| 137 | 7.71 | 6.63 | 7.71 | 7.23 | 6.42 | 7.23 |
| 138 | 7.74 | 6.63 | 7.74 | 7.23 | 6.45 | 7.23 |
| 139 | 7.74 | 6.66 | 7.74 | 7.26 | 6.45 | 7.26 |
| 140 | 7.77 | 6.63 | 7.77 | 7.29 | 6.42 | 7.29 |
| 141 | 7.77 | 6.66 | 7.77 | 7.29 | 6.45 | 7.29 |
| 142 | 7.80 | 6.63 | 7.80 | 7.32 | 6.42 | 7.32 |
| 143 | 7.83 | 6.66 | 7.83 | 7.32 | 6.45 | 7.32 |
| 144 | 7.83 | 6.66 | 7.83 | 7.32 | 6.45 | 7.32 |
| 145 | 7.86 | 6.66 | 7.86 | 7.35 | 6.45 | 7.35 |
| 146 | 7.89 | 6.66 | 7.89 | 7.38 | 6.45 | 7.38 |
| 147 | 7.92 | 6.69 | 7.92 | 7.38 | 6.45 | 7.38 |
| 148 | 7.95 | 6.66 | 7.95 | 7.41 | 6.45 | 7.41 |
| 149 | 7.98 | 6.69 | 7.98 | 7.41 | 6.45 | 7.41 |
| 150 | 7.98 | 6.69 | 7.98 | 7.41 | 6.45 | 7.41 |
| 151 | 8.01 | 6.69 | 8.01 | 7.44 | 6.45 | 7.44 |
| 152 | 8.04 | 6.69 | 8.04 | 7.47 | 6.45 | 7.47 |
| 153 | 8.07 | 6.69 | 8.07 | 7.47 | 6.45 | 7.47 |
| 154 | 8.10 | 6.69 | 8.10 | 7.50 | 6.45 | 7.50 |
| 155 | 8.13 | 6.72 | 8.13 | 7.50 | 6.48 | 7.50 |
| 156 | 8.13 | 6.69 | 8.13 | 7.50 | 6.45 | 7.50 |
| 157 | 8.16 | 6.72 | 8.16 | 7.56 | 6.48 | 7.56 |
| 158 | 8.19 | 6.72 | 8.19 | 7.56 | 6.45 | 7.56 |
| 159 | 8.22 | 6.72 | 8.22 | 7.56 | 6.45 | 7.56 |
|  |  |  |  |  |  |  |


| 160 | 8.25 | 6.72 | 8.25 | 7.59 | 6.48 | 7.59 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 161 | 8.28 | 6.75 | 8.28 | 7.62 | 6.45 | 7.62 |
| 162 | 8.31 | 6.72 | 8.31 | 7.65 | 6.48 | 7.65 |
| 163 | 8.31 | 6.75 | 8.31 | 7.65 | 6.48 | 7.65 |
| 164 | 8.34 | 6.72 | 8.34 | 7.65 | 6.48 | 7.65 |
| 165 | 8.37 | 6.75 | 8.37 | 7.68 | 6.48 | 7.68 |
| 166 | 8.40 | 6.75 | 8.40 | 7.71 | 6.48 | 7.71 |
| 167 | 8.43 | 6.75 | 8.43 | 7.74 | 6.48 | 7.74 |
| 168 | 8.46 | 6.75 | 8.46 | 7.74 | 6.48 | 7.74 |
| 169 | 8.49 | 6.78 | 8.49 | 7.74 | 6.48 | 7.74 |
| 170 | 8.52 | 6.78 | 8.52 | 7.80 | 6.48 | 7.80 |
| 171 | 8.55 | 6.78 | 8.55 | 7.80 | 6.48 | 7.80 |
| 172 | 8.55 | 6.81 | 8.55 | 7.83 | 6.48 | 7.83 |
| 173 | 8.58 | 6.78 | 8.58 | 7.86 | 6.48 | 7.86 |
| 174 | 8.61 | 6.81 | 8.61 | 7.86 | 6.51 | 7.86 |
| 175 | 8.67 | 6.81 | 8.67 | 7.89 | 6.48 | 7.89 |
| 176 | 8.70 | 6.81 | 8.70 | 7.89 | 6.51 | 7.89 |
| 177 | 8.73 | 6.81 | 8.73 | 7.95 | 6.48 | 7.95 |
| 178 | 8.76 | 6.81 | 8.76 | 7.95 | 6.48 | 7.95 |
| 179 | 8.79 | 6.81 | 8.79 | 7.98 | 6.51 | 7.98 |
| 180 | 8.82 | 6.84 | 8.82 | 8.01 | 6.48 | 8.01 |
| 181 | 8.85 | 6.81 | 8.85 | 8.01 | 6.51 | 8.01 |
| 182 | 8.88 | 6.84 | 8.88 | 8.04 | 6.51 | 8.04 |
| 183 | 8.91 | 6.84 | 8.91 | 8.04 | 6.48 | 8.04 |
| 184 | 8.94 | 6.84 | 8.94 | 8.10 | 6.51 | 8.10 |
| 185 | 9.00 | 6.84 | 9.00 | 8.10 | 6.51 | 8.10 |
| 186 | 9.03 | 6.87 | 9.03 | 8.13 | 6.51 | 8.13 |
| 187 | 9.06 | 6.84 | 9.06 | 8.16 | 6.51 | 8.16 |
| 188 | 9.09 | 6.87 | 9.09 | 8.19 | 6.51 | 8.19 |
| 189 | 9.12 | 6.90 | 9.12 | 8.19 | 6.51 | 8.19 |
| 190 | 9.15 | 6.87 | 9.15 | 8.25 | 6.51 | 8.25 |
| 191 | 9.18 | 6.90 | 9.18 | 8.25 | 6.51 | 8.25 |
| 192 | 9.21 | 6.90 | 9.21 | 8.28 | 6.51 | 8.28 |
| 193 | 9.27 | 6.90 | 9.27 | 8.31 | 6.54 | 8.31 |
| 194 | 9.30 | 6.90 | 9.30 | 8.34 | 6.51 | 8.34 |
| 195 | 9.33 | 6.93 | 9.33 | 8.34 | 6.54 | 8.34 |
| 196 | 9.36 | 6.90 | 9.36 | 8.40 | 6.54 | 8.40 |
| 197 | 9.39 | 6.93 | 9.39 | 8.40 | 6.51 | 8.40 |
| 198 | 9.42 | 6.93 | 9.42 | 8.46 | 6.54 | 8.46 |
| 199 | 9.48 | 6.93 | 9.48 | 8.46 | 6.51 | 8.46 |
| 200 | 9.51 | 6.93 | 9.51 | 8.49 | 6.54 | 8.49 |
| 201 | 9.54 | 6.96 | 9.54 | 8.52 | 6.54 | 8.52 |
| 202 | 9.57 | 6.96 | 9.57 | 8.55 | 6.51 | 8.55 |
| 203 | 9.60 | 6.96 | 9.60 | 8.61 | 6.54 | 8.61 |
| 204 | 9.66 | 6.99 | 9.66 | 8.61 | 6.54 | 8.61 |
| 205 | 9.69 | 6.96 | 9.69 | 8.64 | 6.54 | 8.64 |
| 206 | 9.72 | 6.99 | 9.72 | 8.67 | 6.54 | 8.67 |
|  |  |  |  |  |  |  |


| 207 | 9.75 | 6.99 | 9.75 | 8.70 | 6.57 | 8.70 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 208 | 9.78 | 6.99 | 9.78 | 8.73 | 6.54 | 8.73 |
| 209 | 9.84 | 6.99 | 9.84 | 8.76 | 6.54 | 8.76 |
| 210 | 9.87 | 7.02 | 9.87 | 8.82 | 6.54 | 8.82 |
| 211 | 9.93 | 6.99 | 9.93 | 8.82 | 6.54 | 8.82 |
| 212 | 9.96 | 7.02 | 9.96 | 8.85 | 6.57 | 8.85 |
| 213 | 10.00 | 7.05 | 10.00 | 8.88 | 6.54 | 8.88 |
| 214 | 10.06 | 7.02 | 10.06 | 8.91 | 6.57 | 8.91 |
| 215 | 10.09 | 7.05 | 10.09 | 8.97 | 6.57 | 8.97 |
| 216 | 10.12 | 7.05 | 10.12 | 8.97 | 6.54 | 8.97 |
| 217 | 10.15 | 7.05 | 10.15 | 9.03 | 6.57 | 9.03 |
| 218 | 10.21 | 7.05 | 10.21 | 9.06 | 6.57 | 9.06 |
| 219 | 10.24 | 7.08 | 10.24 | 9.09 | 6.57 | 9.09 |
| 220 | 10.30 | 7.08 | 10.30 | 9.12 | 6.57 | 9.12 |
| 221 | 10.33 | 7.08 | 10.33 | 9.18 | 6.57 | 9.18 |
| 222 | 10.39 | 7.11 | 10.39 | 9.18 | 6.57 | 9.18 |
| 223 | 10.42 | 7.08 | 10.42 | 9.24 | 6.57 | 9.24 |
| 224 | 10.45 | 7.11 | 10.45 | 9.27 | 6.57 | 9.27 |
| 225 | 10.48 | 7.11 | 10.48 | 9.30 | 6.57 | 9.30 |
| 226 | 10.54 | 7.11 | 10.54 | 9.33 | 6.60 | 9.33 |
| 227 | 10.60 | 7.11 | 10.60 | 9.39 | 6.57 | 9.39 |
| 228 | 10.63 | 7.14 | 10.63 | 9.42 | 6.60 | 9.42 |
| 229 | 10.69 | 7.14 | 10.69 | 9.45 | 6.60 | 9.45 |
| 230 | 10.72 | 7.14 | 10.72 | 9.51 | 6.57 | 9.51 |
| 231 | 10.75 | 7.17 | 10.75 | 9.54 | 6.60 | 9.54 |
| 232 | 10.78 | 7.14 | 10.78 | 9.57 | 6.60 | 9.57 |
| 233 | 10.84 | 7.17 | 10.84 | 9.60 | 6.60 | 9.60 |
| 234 | 10.90 | 7.20 | 10.90 | 9.66 | 6.60 | 9.66 |
| 235 | 10.93 | 7.20 | 10.93 | 9.72 | 6.57 | 9.72 |
| 236 | 10.96 | 7.20 | 10.96 | 9.72 | 6.60 | 9.72 |
| 237 | 11.02 | 7.20 | 11.02 | 9.78 | 6.60 | 9.78 |
| 238 | 11.05 | 7.23 | 11.05 | 9.84 | 6.60 | 9.84 |
| 239 | 11.11 | 7.20 | 11.11 | 9.87 | 6.60 | 9.87 |
| 240 | 11.17 | 7.23 | 11.17 | 9.93 | 6.63 | 9.93 |
| 241 | 11.20 | 7.26 | 11.20 | 9.93 | 6.60 | 9.93 |
| 242 | 11.23 | 7.23 | 11.23 | 10.00 | 6.60 | 10.00 |
| 243 | 11.29 | 7.26 | 11.29 | 10.06 | 6.63 | 10.06 |
| 244 | 11.32 | 7.26 | 11.32 | 10.09 | 6.60 | 10.09 |
| 245 | 11.38 | 7.26 | 11.38 | 10.15 | 6.63 | 10.15 |
| 246 | 11.41 | 7.29 | 11.41 | 10.18 | 6.63 | 10.18 |
| 247 | 11.47 | 7.29 | 11.47 | 10.21 | 6.63 | 10.21 |
| 248 | 11.50 | 7.32 | 11.50 | 10.27 | 6.63 | 10.27 |
| 249 | 11.53 | 7.29 | 11.53 | 10.33 | 6.60 | 10.33 |
| 250 | 11.62 | 7.32 | 11.62 | 10.39 | 6.63 | 10.39 |
| 251 | 11.65 | 7.35 | 11.65 | 10.42 | 6.63 | 10.42 |
| 252 | 11.68 | 7.32 | 11.68 | 10.48 | 6.63 | 10.48 |
| 253 | 11.74 | 7.35 | 11.74 | 10.54 | 6.63 | 10.54 |
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| 254 | 11.77 | 7.35 | 11.77 | 10.54 | 6.66 | 10.54 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 255 | 11.80 | 7.38 | 11.80 | 10.60 | 6.63 | 10.60 |
| 256 | 11.89 | 7.38 | 11.89 | 10.66 | 6.63 | 10.66 |
| 257 | 11.92 | 7.38 | 11.92 | 10.72 | 6.66 | 10.72 |
| 258 | 11.95 | 7.41 | 11.95 | 10.75 | 6.63 | 10.75 |
| 259 | 12.01 | 7.38 | 12.01 | 10.81 | 6.66 | 10.81 |
| 260 | 12.04 | 7.41 | 12.04 | 10.87 | 6.66 | 10.87 |
| 261 | 12.10 | 7.44 | 12.10 | 10.93 | 6.66 | 10.93 |
| 262 | 12.13 | 7.41 | 12.13 | 10.99 | 6.66 | 10.99 |
| 263 | 12.19 | 7.44 | 12.19 | 11.02 | 6.66 | 11.02 |
| 264 | 12.22 | 7.44 | 12.22 | 11.08 | 6.66 | 11.08 |
| 265 | 12.25 | 7.47 | 12.25 | 11.14 | 6.66 | 11.14 |
| 266 | 12.31 | 7.47 | 12.31 | 11.20 | 6.69 | 11.20 |
| 267 | 12.37 | 7.47 | 12.37 | 11.23 | 6.66 | 11.23 |
| 268 | 12.40 | 7.50 | 12.40 | 11.29 | 6.69 | 11.29 |
| 269 | 12.46 | 7.53 | 12.46 | 11.35 | 6.66 | 11.35 |
| 270 | 12.49 | 7.50 | 12.49 | 11.41 | 6.66 | 11.41 |
| 271 | 12.52 | 7.53 | 12.52 | 11.47 | 6.69 | 11.47 |
| 272 | 12.58 | 7.53 | 12.58 | 11.53 | 6.66 | 11.53 |
| 273 | 12.64 | 7.53 | 12.64 | 11.56 | 6.69 | 11.56 |
| 274 | 12.67 | 7.56 | 12.67 | 11.62 | 6.69 | 11.62 |
| 275 | 12.73 | 7.56 | 12.73 | 11.68 | 6.69 | 11.68 |
| 276 | 12.76 | 7.59 | 12.76 | 11.74 | 6.69 | 11.74 |
| 277 | 12.79 | 7.59 | 12.79 | 11.80 | 6.69 | 11.80 |
| 278 | 12.85 | 7.59 | 12.85 | 11.86 | 6.69 | 11.86 |
| 279 | 12.88 | 7.62 | 12.88 | 11.89 | 6.69 | 11.89 |
| 280 | 12.94 | 7.62 | 12.94 | 11.95 | 6.72 | 11.95 |
| 281 | 13.00 | 7.62 | 13.00 | 12.01 | 6.69 | 12.01 |
| 282 | 13.03 | 7.65 | 13.03 | 12.07 | 6.72 | 12.07 |
| 283 | 13.06 | 7.65 | 13.06 | 12.13 | 6.72 | 12.13 |
| 284 | 13.12 | 7.68 | 13.12 | 12.19 | 6.72 | 12.19 |
| 285 | 13.15 | 7.68 | 13.15 | 12.25 | 6.72 | 12.25 |
| 286 | 13.18 | 7.68 | 13.18 | 12.28 | 6.72 | 12.28 |
| 287 | 13.27 | 7.71 | 13.27 | 12.37 | 6.72 | 12.37 |
| 288 | 13.30 | 7.74 | 13.30 | 12.43 | 6.72 | 12.43 |
| 289 | 13.33 | 7.71 | 13.33 | 12.49 | 6.75 | 12.49 |
| 290 | 13.39 | 7.74 | 13.39 | 12.55 | 6.72 | 12.55 |
| 291 | 13.42 | 7.77 | 13.42 | 12.61 | 6.75 | 12.61 |
| 292 | 13.45 | 7.77 | 13.45 | 12.67 | 6.72 | 12.67 |
| 293 | 13.51 | 7.77 | 13.51 | 12.70 | 6.72 | 12.70 |
| 294 | 13.54 | 7.80 | 13.54 | 12.76 | 6.75 | 12.76 |
| 295 | 13.57 | 7.80 | 13.57 | 12.82 | 6.72 | 12.82 |
| 296 | 13.63 | 7.83 | 13.63 | 12.88 | 6.75 | 12.88 |
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